

# 1 Bayesian operational modal analysis of closely spaced 2 modes for monitoring wind turbines

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## 7 **Abstract**

8 In this study, the applicability of Bayesian operational modal analysis (BAY-  
9 OMA) to an operating onshore concrete-steel hybrid wind turbine tower is  
10 investigated. The results of the identification then provide reliable parameters  
11 for the structural health monitoring (SHM) of the tower.

12 In the context of wind turbines, typical assumptions of linear time-invariant  
13 OMA methods are violated, so the validity of the identification uncertainties of  
14 BAYOMA is not necessarily given. In addition, closely spaced modes occur, for  
15 which the mode shape in particular is subject to high uncertainty. It can be  
16 stated, that the main part of the mode shape uncertainty corresponds to the  
17 alignment of these in the mode subspace.

18 Due of these challenges, mode shapes are generally not taken into account  
19 when monitoring wind turbine towers. In order to include the mode shape in  
20 SHM scheme, the second-order modal assurance criterion (S2MAC) is applied  
21 in this study. This metric is able to eliminate the alignment uncertainty by  
22 comparing the mode shape with a mode subspace.

23 Besides mode shapes, the reliability of natural frequencies and damping can  
24 also be better quantified by knowing the identification uncertainty. This finally  
25 enables a well-founded selection of suitable monitoring parameters for the  
26 future application of SHM for wind turbines.

27 *Keywords:* BAYOMA, wind turbine tower, structural health monitoring,  
28 uncertainty quantification, closely spaced modes

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## 29 **1. Introduction**

30 Wind energy already accounts for the largest share of renewable electricity  
31 generation in the European Union (EU). In 2018, wind energy accounted for  
32 18.4% of the electricity generation capacity in the EU, with an installed capacity  
33 of 170 Gigawatt (GW) onshore and 19 GW offshore [1]. As in many engineer-  
34 ing disciplines, efficient operation and maintenance also play a major role in  
35 the field of wind turbines. Consequently, there is a great motivation to im-  
36 plement effective monitoring strategies in order to reduce maintenance costs  
37 and increase safety at the same time [2]. In the field of civil engineering, the  
38 associated monitoring concept is referred to as *Structural Health Monitoring*  
39 (SHM). In this context, a distinction is generally made between model-based  
40 SHM and data-based SHM. Data-based SHM is currently considered the pre-  
41 dominant approach [3]. To apply data-based methods, a suitable measurement  
42 concept is crucial. A global monitoring approach is often used due to a more

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43 economic measurement concept compared to local approaches. Here, a small  
44 number of sensors is used to determine information about the condition of  
45 the whole structure in terms of structural dynamics. Based on the measured  
46 system response data, monitoring parameters (MP) are extracted using feature  
47 extraction techniques. From these parameters, a subset of parameters is deter-  
48 mined that can distinguish between a damaged and an undamaged state of the  
49 structure [4]. In this context, operational modal analysis (OMA) methods are  
50 commonly used to identify modal parameters as MPs. Typically, the result of  
51 the identification of modal parameters includes natural frequencies, damping  
52 and mode shapes and does not require measurements of the excitation forces.  
53 In the recent years, OMA methods, like *Bayesian Operational Modal Analy-*  
54 *sis* (BAYOMA) [5], were developed to not only identify the modal parameters  
55 but also their uncertainties.

56 Various OMA methods have been successfully used in recent years for mon-  
57 itoring the support structures of wind turbines. In particular, the covariance-  
58 based *Stochastic Subspace Identification* (SSI) [6, 7, 8], the *poly-reference*  
59 *Least Squares Complex Frequency* (pLSCF) [7, 8] and the *Frequency Domain*  
60 *Decomposition* (FDD) [9, 10] are to be mentioned here. The aim of this work is  
61 to investigate the suitability of the relatively recent method BAYOMA for mon-  
62 itoring of an existing hybrid tower of an onshore wind turbine. Challenges in  
63 this specific application include the identification of closely spaced modes, har-  
64 monic excitation, a short evaluation time relative to the oscillation period, high  
65 damping, and non-stationarity due to environmental and operational conditions

66 (EOC).

67 The low-frequency system dynamics are challenging, because the measure-  
68 ment chain must be designed and calibrated for the low-frequency range [11].

69 In addition, this makes the identification of the modal parameters much more  
70 uncertain [12]. The Identification of a wind turbine tower is further complicated

71 by the fact that the damping of the fore-aft mode (FA) at higher rotor speeds  
72 is greater than that of the side-to-side mode (SS) [9]. If the evaluation time is

73 extended beyond the commonly used 10 minutes to improve the identification  
74 accuracy, there is a risk that the identification will become less reliable due

75 to the instationarities caused by varying EOCs. This problem can be solved  
76 using time-varying systems, like time-varying autoregressive moving average

77 models (TV ARMA) [13]. However, linear OMA methods are commonly used  
78 for vibration-based monitoring, which assume a time-invariant system under

79 white noise excitation. These identification procedures were found to be ro-  
80 bust even when the assumption of time invariance was violated, as found by

81 Brownjohn et al. [14], who applied BAYOMA to offshore lighthouses. Another  
82 challenge is the harmonic excitation, which can lead to distortion of the natural

83 frequencies [6]. Possible approaches in the context of monitoring wind turbine  
84 towers for example do not consider the identified natural frequencies in the

85 range of higher harmonics of the rotor [15], or use cluster analysis to separate  
86 natural frequencies from harmonics [16].

87 In tower structures, it is common to deal with closely spaced modes which are  
88 challenging to identify, especially regarding the mode shapes. Au et al. [12]



89 show that the largest uncertainty in the case of closely spaced modes occurs  
90 for the identification of the mode shapes. This uncertainty can be divided  
91 into two parts. The first part is the uncertainty of the mode subspace (MSS)  
92 spanned by the dominant vibration shapes. This uncertainty is similar to the  
93 uncertainty of mode shapes in the well-separated case, which depends mainly  
94 on the noise of the measurement chain. Hence, in case of low-noise data, the  
95 MSS can be identified very reliably. The second part of the uncertainty of the  
96 mode shape is the alignment of the mode in the MSS. The uncertainty of the  
97 alignment identification increases significantly with the increase in closeness  
98 of the frequencies [17]. Therefore, an extension of the well-known modal  
99 assurance criterion (MAC) in form of the subspace of order 2 MAC (S2MAC)  
100 [18] was developed, which compares a mode shape with a subspace. This  
101 metric can provide in case of closely spaced modes less uncertain results than  
102 the classical MAC, because it eliminates uncertainty in the alignment. This  
103 allows detection of system changes for symmetrical tower structures based on  
104 mode shapes [19, 17].

105 In this study, BAYOMA is used as it identifies their uncertainties in addition to  
106 the modal parameters, and gives good results in the context of closely spaced  
107 bending modes of tower structures [14, 17]. The aim of this study is to investi-  
108 gate the applicability of BAYOMA with the associated identification uncertainties  
109 on a hybrid tower of an onshore wind turbine under operating conditions in  
110 order to obtain meaningful monitoring parameters for structural health moni-  
111 toring. The structure of the work is as follows: Section 2 introduces the theory

112 used in the following chapters. Section 3 describes the wind turbine tower  
113 under investigation and its dynamics are analysed in more detail, taking the  
114 identification uncertainty into account. Finally, in Section 4, the study is sum-  
115 marised and an outlook is given.

## 116 **2. Theory**

117 This section explains shortly BAYOMA and the metrics for comparing mode  
118 shapes of closely spaced modes. In addition, the mode tracking of the modes  
119 of a wind turbine tower under operation is presented.

### 120 *2.1. Bayesian operational modal analysis*

121 In this study, the natural frequencies and mode shapes are identified with the  
122 frequency domain method BAYOMA [5, 20]. The basis of BAYOMA is the  
123 discrete Fourier transform (DFT) of a Gaussian distributed signal. Assuming  
124 a long measurement time and a high sampling rate, a DFT of the individual  
125 frequency base point is statistically independent of all other base points and  
126 also Gaussian distributed [21]. By assuming an equally distributed prior of the  
127 modal parameters, the likelihood becomes proportional to the posterior.

128 The likelihood of the DFT is therefore a multivariate Gaussian distribution. The  
129 related covariance matrix is the expected power spectral density matrix of  $m$   
130 dominating modes. In case of several closely spaced modes in a considered  
131 frequency range, the variables to be identified increase significantly due to the  
132 number of mode shapes. To reduce numerical complexity, the mode subspace

133 (MSS) is first identified and then the alignment of the modes in the subspace  
 134 as well as the associated natural frequencies and damping are identified. The  
 135 MSS, noted as  $\Psi$ , is a subspace spanned by the  $m$  dominating vibration shapes  
 136 in the considered frequency range. Assuming real mode shapes, these are  
 137 formed from the  $m$  largest eigenvectors of the summed real spectral matrix  
 138 in the considered frequency range. Instead of the entire mode shape, the  
 139 identification only needs to determine the angles  $\beta$  of the transformation matrix  
 140  $T$  corresponding to the orientation of the mode within the mode subspace. For  
 141  $m = 2$ , this can be defined as

$$\Phi = \Psi_{1,2} T(\beta_1, \beta_2) = \Psi_{1,2} \begin{bmatrix} \cos(\beta_1) & \cos(\beta_2) \\ \sin(\beta_1) & \sin(\beta_2) \end{bmatrix}. \quad (1)$$

142 This provides the expected value  $\mathbb{E}$  of the theoretical power spectral density  
 143 matrix of two dominating modes

$$\mathbb{E}_k(\Theta) = \Psi_{1,2} T H_k (\Psi_{1,2} T)^T + S_e \Psi \Psi^T, \quad (2)$$

144 where  $H_k$  is a diagonal matrix containing the two theoretical power spectral  
 145 densities of equivalent one-mass oscillators for the frequency support point  $k$ .  
 146 The optimisation parameters  $\Theta$  are the natural frequencies, modal damping,  
 147 modal force, the angles of the transformation matrix as well as the model error  
 148  $S_e$ . The identification of the most probable values of  $\Theta$  for a specified frequency

149 range is preformed by minimising the negative log likelihood function  $L(\Theta)$

$$L(\Theta) = n_c N_f \ln \pi + \sum_{k=1}^{N_f} \ln |\mathbb{E}_k(\Theta)| + \sum_{k=1}^{N_f} \mathcal{F}_k^* \mathbb{E}_k(\Theta)^{-1} \mathcal{F}_k, \quad (3)$$

150 where  $N_f$  is the number of considered frequency points and  $\mathcal{F}$  is the DFT of  
 151 the measured signal. The covariance matrix of the Gaussian approximation of  
 152 the posterior distribution is calculated with the inverse Hessian matrix of the  
 153 negative log likelihood function at the most probable values. In a subsequent  
 154 step, the MSS can be adjusted using a Newton iteration [20].

## 155 2.2. Metrics for mode shapes

156 For almost rotationally symmetric tower structures, closely spaced modes oc-  
 157 cur for the bending modes. For such structures, previous investigations have  
 158 shown that the mode shapes have much higher associated identification uncer-  
 159 tainty than in the case of well-separated modes [12, 19]. Most of the uncertainty  
 160 is in the alignment of the mode shape in the MSS, so the widely used Modal As-  
 161 surance Criterion (MAC) [22] to compare two mode shapes  $\boldsymbol{\varphi}_j$  and  $\boldsymbol{\varphi}_k$ , defined  
 162 as

$$\text{MAC}_{j,k} = \frac{|\boldsymbol{\varphi}_j^H \boldsymbol{\varphi}_k|^2}{\boldsymbol{\varphi}_j^H \boldsymbol{\varphi}_j \boldsymbol{\varphi}_k^H \boldsymbol{\varphi}_k}, \quad (4)$$

163 becomes very uncertain as well. This is visualised by the identification un-  
 164 certainty of an exemplary closely spaced mode shape using a Monte Carlo  
 165 simulation in Figure 1A, where the scatter is evident. Moreover, no Gaussian  
 166 distribution can be assumed when the MAC values approach one. The assump-

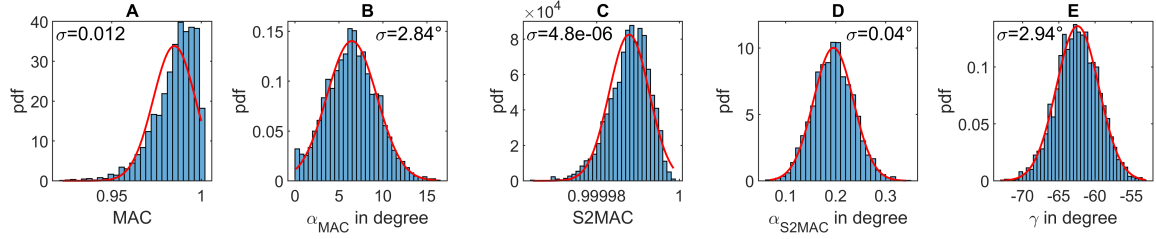


Figure 1: Histograms with normal distribution fits and the standard deviation  $\sigma$  of the MAC,  $\alpha_{MAC}$ , S2MAC,  $\alpha_{S2MAC}$  and the directional angle  $\gamma$  of one exemplary mode shape of a closely spaced bending mode pair, normalised according to the probability density functions (pdf). The determination is performed using the covariance matrix of the mode shape and the Monte Carlo method with 3000 samples.

167 tion of a beta distribution is better suited to modelling the MAC distribution  
 168 [17].

169 To eliminate the alignment uncertainty, the subspace of order 2 Modal Assur-  
 170 ance Criterion (S2MAC) was developed [18]. The S2MAC calculates the best  
 171 MAC between the mode shape  $\varphi_i$  and the MSS spanned by two vibration shapes  
 172 vectors  $\psi_j$  and  $\psi_k$ . In the case of normalised real mode shapes of length one,  
 173 the S2MAC is defined as

$$\text{S2MAC}_{i,jk} = \frac{(\varphi_i^T \psi_j)^2 - 2(\varphi_i^T \psi_j)(\psi_j^T \psi_k)(\varphi_i^T \psi_k) + (\varphi_i^T \psi_k)^2}{1 - (\psi_j^T \psi_k)^2}. \quad (5)$$

174 Figure 1C shows that the scattering is significantly reduced compared to the  
 175 regular MAC and the distribution is closer to a Gaussian distribution. However,  
 176 a slight skewness of the distribution is still present. The MAC and S2MAC  
 177 are relatively insensitive to small changes of the mode shape relative to the  
 178 reference shapes. Since both metrics represents a squared scalar product of

179 vectors normalised to one, the angles  $\alpha_{MAC}$  and  $\alpha_{S2MAC}$  can be derived

$$\alpha_{MAC} = \arccos(\sqrt{MAC}). \quad (6)$$

180 The angle of the MAC is simply the angle between the two mode shapes, for  
181 the S2MAC it is the smallest angle between the mode shape and the MSS. Due  
182 to this transformation,  $\alpha_{MAC}$  and  $\alpha_{S2MAC}$  become Gaussian distributed, which is  
183 shown in Figure 1B and 1D. In Figure 1B, the  $\alpha_{MAC}$  illustrates that the angle  
184 representation can contain a deviation from the Gaussian distribution close to  
185 zero, due to the fact that the angles are constrained to be larger than 0. This  
186 error occurs in the case of large uncertainty and mean values close to 0. In  
187 the context of wind turbines, this can occur especially with the significantly less  
188 reliable  $\alpha_{MAC}$ .

189 The alignment uncertainty can be approximated by the directional angle  $\gamma$  in  
190 the case of a tower structure and a same sensor setup at all measurement levels  
191 in both spatial directions analogous to the calculation of the mean phase [23],  
192 as shown in [17]

$$\gamma = \arctan\left(\frac{-V_{12}}{V_{22}}\right) \text{ with } USV^T = [\boldsymbol{\varphi}_x \ \boldsymbol{\varphi}_y], \quad (7)$$

193 where  $USV^T$  is the singular value decomposition,  $\boldsymbol{\varphi}_x$  are the entries of the  
194 mode shape in  $x$ -direction, and  $\boldsymbol{\varphi}_y$  are the entries of the mode shape in  $y$ -  
195 direction.  $V_{12}$  and  $V_{22}$  are the corresponding elements of the matrix  $V$ . The  
196 distribution of the direction angle in Figure 1E demonstrates clearly that it

197 can be assumed as Gaussian. It is remarkable that the standard deviation of  
198 the  $\alpha_{MAC}$  and the direction angle are quite similar. This is an indication that  
199 the alignment uncertainty of the mode in the MSS is well described by the  
200 directional angle  $\gamma$  in the case of bending modes of tower structures. In this  
201 study, the uncertainties of the mode shape metrics are determined with a 3000  
202 sample Monte Carlo simulation, which takes into account the covariance matrix  
203 of the mode shape identification. The distribution of the mode shape metrics  
204  $\alpha_{MAC}$  ,  $\alpha_{S2MAC}$  and  $\gamma$  assumed to be Gaussian despite the possible small error.

### 205 *2.3. Mode tracking of closely spaced modes*

206 In the case of changing modal parameters caused by varying EOCs or mechan-  
207 ical changes, mode tracking becomes a challenge. Here, the identified natural  
208 frequencies and mode shapes are compared with reference frequencies or ref-  
209 erence shapes. As demonstrated in the previous chapter, an assignment of the  
210 mode shapes in the presence of closely spaced modes is associated with great  
211 uncertainties. In case of support structures of wind turbines, this is further  
212 complicated, because the mode alignment changes along with a changing na-  
213 celle positions. A typical approach for this application is to rotate the reference  
214 mode shape depending on the nacelle position [16]. Subsequently, the rotated  
215 reference mode shape can be compared to the identified mode with the MAC,  
216 such that it becomes insensitive to the nacelle angle. For this study, a similar  
217 procedure was used, which is shown in Figure 2. First, the modal parameters  
218 and the associated uncertainties are identified from the acceleration measure-  
219 ment data using BAYOMA. Since BAYOMA is a non-parametric identification

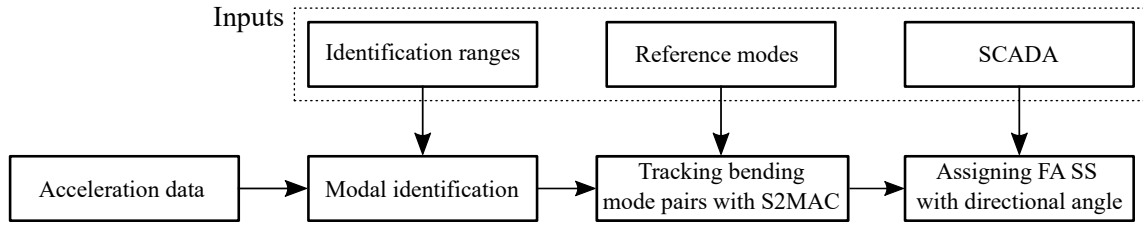


Figure 2: Sequence of identification and mode tracking used to monitor the support structure of a wind turbine.

220 method, the identification ranges and the number of modes within a frequency  
 221 range are required as prerequisite information. These informations can be  
 222 provided either automatically, according to Brincker et al. [24] or manually. In  
 223 this work, the identification ranges are set manually. This has the advantage  
 224 that only the modes of interest are identified. However, care must be taken to  
 225 ensure that these ranges are sufficiently large to include the full range of vari-  
 226 ability and that a verification of the identification results is carried out. To verify  
 227 that two different modes have been identified, it is required that the maximum  
 228 MAC of the two closely spaced mode shapes does not exceed 0.5 to obtain  
 229 two different modes and that the identified natural frequencies are within the  
 230 identification range.

231 In a further part of the verification, the S2MAC is used to check whether  
 232 the identified mode matches the previously determined reference MSS. This  
 233 has the advantage that the alignment of the mode shape in the mode subspace,  
 234 which is the main uncertainty of the mode shape for closely spaced modes, does  
 235 not influence the bending mode pair tracking process. In addition, the influence  
 236 of the nacelle position on the bending mode pair tracking can be eliminated,



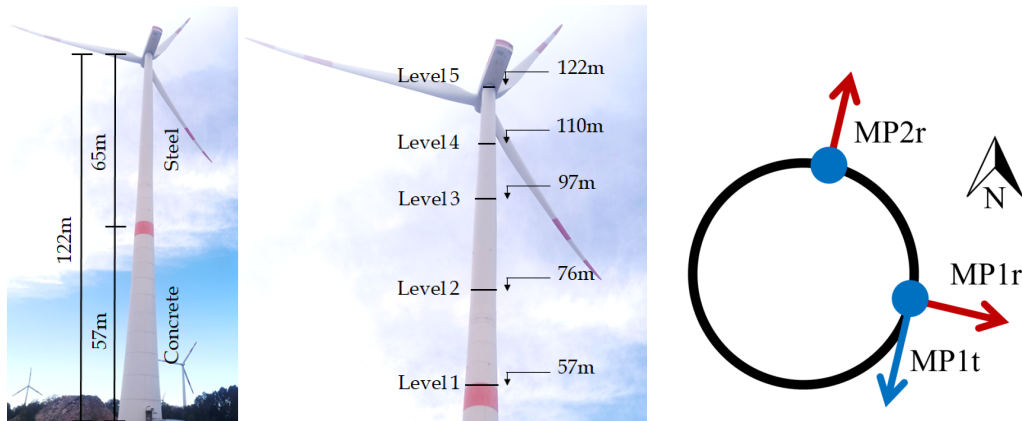


Figure 3: Sensor setup on the steel-concrete hybrid tower of a wind turbine. MP2 is aligned in  $10^\circ$  North and MP1 in  $100^\circ$  East.

237 which is advantageous in case of non-synchronous aggregated *Supervisory*  
 238 *Control And Data Acquisition* (SCADA). The assignment of an identified mode  
 239 to a bending mode pair is done when the S2MAC is greater than 0.8. Lower  
 240 values of S2MAC are considered to be misidentifications.

241 However, for the distinction of the modes according to FA and SS within a  
 242 bending mode pair, the nacelle position is required. This is achieved by classi-  
 243 fying the mode whose directional angle  $\gamma$  is closest to the nacelle position as  
 244 the FA-mode. The other mode is correspondingly assigned as the SS-mode.

### 245 3. Investigated hybrid concrete steel tower of a wind turbine

246 In this study, a hybrid concrete and steel tower of an 3.4 MW onshore wind  
 247 turbine is investigated, which is shown in Figure 3. The first 57 m of the 122 m  
 248 high tower consist of prestressed segmented concrete rings. The upper part is  
 249 composed of steel tubes. The rated rotor speed of 14 rpm is reached at a wind

250 speed of about 10 m/s. For the wind turbine under investigation, the main wind  
251 direction is West. This section first briefly describes the measurement setup.  
252 Subsequently, the dynamics of the tower is investigated on the basis of the first  
253 and fourth bending mode pairs.

### 254 *3.1. Measurement setup*

255 A measuring system is installed in an existing wind turbine. Because of lim-  
256 ited accessibility, the five measuring levels coincide with the platforms of the  
257 towers. On each level, three *Integrated Electronics Piezo-Electric* (IEPE) ac-  
258 celerometers are installed. Two sensors measure in the radial direction of the  
259 tower, with a 90° angle to each other (MP1r and MP2r). An additional sensor of  
260 tangential direction is attached to one measuring point (MP1t) per measuring  
261 level, as shown in Figure 3. The calibrated IEPE sensors are combined with an  
262 IEPE supply with a cut-off frequency of 0.0106 Hz, enabling the measurement  
263 of acceleration signals without distortion down to 0.05 Hz [11]. The measure-  
264 ment data of all sensors are digitised synchronously with a 24 bit analogue to  
265 digital converter on Level 1 positioned and stored with a sampling rate of 500  
266 Hz on a computer. The aim of this experimental setup is to investigate the dy-  
267 namics of the hybrid tower in operation and to detect possible system changes  
268 over time. For the evaluation in this work, only the two acceleration sensors  
269 of MP1 from each measurement level are used. For a detailed study of the  
270 dynamics of the tower presented in the next section, measurement data sets  
271 from middle of October 2021 to end of September 2022 are used, assuming  
272 enough EOC variation during this period.

Mode pair	$f_0$ FA	$f_0$ SS	identification range	identification rate
1	0.309 Hz	0.297 Hz	0.24 Hz - 0.36 Hz	61.9%
2	1.445 Hz	1.475 Hz	1.25 Hz - 1.7 Hz	40.9%
3	3.144 Hz	3.036 Hz	2.8 Hz - 3.35 Hz	54.5%
4	3.602 Hz	3.788 Hz	3.35 Hz - 4.2 Hz	78.5%

Table 1: Median of the natural frequencies  $f_0$ , identification range and identification rate of the studied bending mode pairs for the selected EOC's in the period from middle of October 2021 to end of September 2022.

### 273 3.2. Dynamic of the tower

274 To use BAYOMA, the identification ranges must be defined a-priori, as de-  
 275 scribed in Section 2.3. In the frequency range up to 5 Hz, four bending mode  
 276 pairs occur. The identification ranges for these mode pairs are listed in Table 1.

The mode shapes of these modes are shown in Figure 4. The bending modes

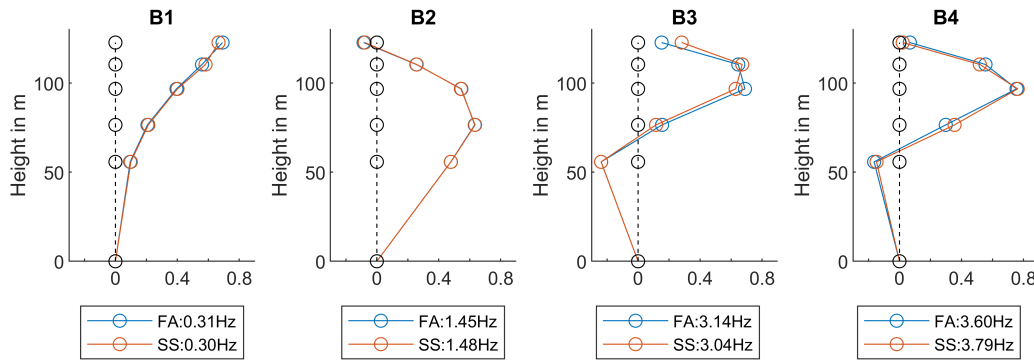


Figure 4: Mode shapes of the four bending mode pairs, identified under operation at a nacelle position of  $270^\circ$ , and rotated in the dominant direction for comparability.

277

278 are similar in FA and SS direction, respectively. The slight deviations may re-  
 279 sult from an asymmetric stiffness distribution around the circumference of the  
 280 tower or the unevenly distributed head mass through the rotor and nacelle.

281 Figure 5 shows the trend of the natural frequencies over time. As generally  
 282 known, natural frequencies change over time due to EOCs. In addition, there

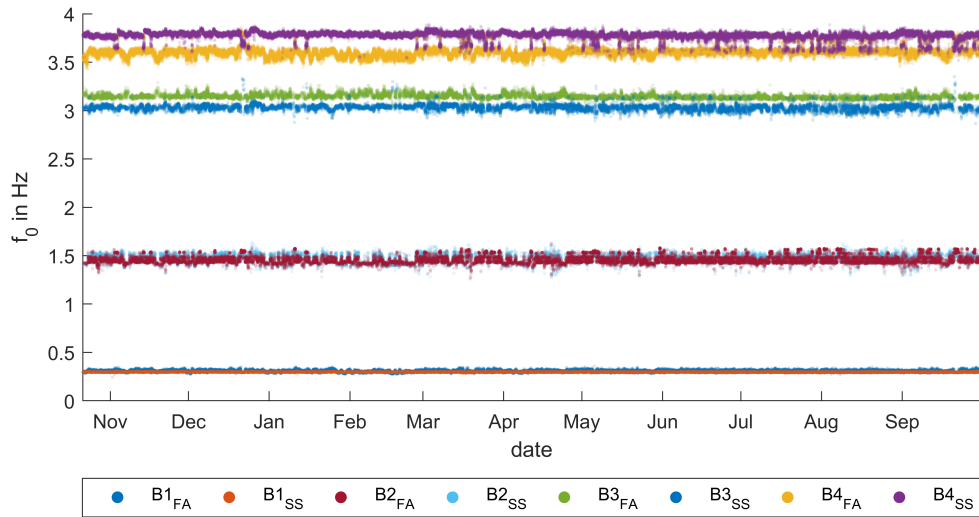


Figure 5: The natural frequencies of the four identified mode pairs plotted over time.

283 appear to be assignment issues, especially with the second and fourth mode  
 284 pair. A better insight is provided by the Campbell diagram in Figure 6, which  
 shows the natural frequencies as a function of rotor speed. The harmonic ex-

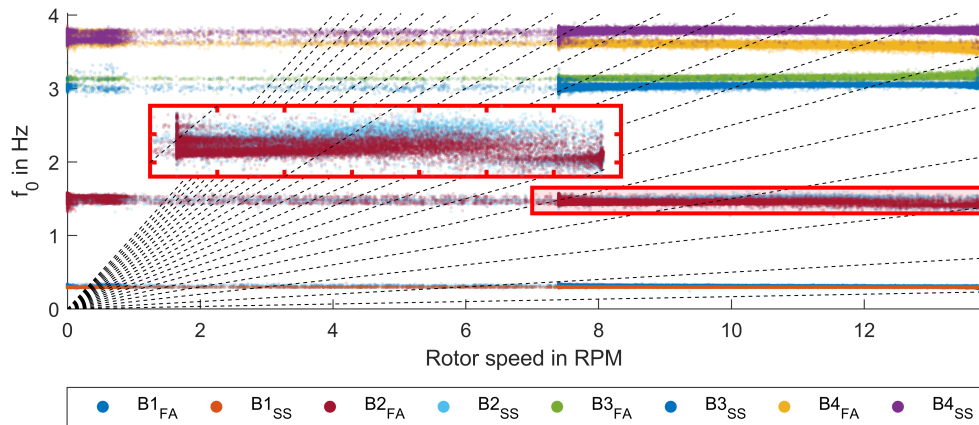


Figure 6: Campbell diagram with the natural frequencies of the four lowest bending mode pairs.

285

286 citation has no relevant influence on the identification of the modal parameters  
 287 as the dashed lines of the higher harmonics of the rotor speed do not correlate

288 with the identified natural frequencies. In addition, the assignment problem of  
 289 the fourth bending mode pair mainly occurs in standstill and start-up condi-  
 290 tions. However, the second bending mode pair around 1.5 Hz scatters strongly  
 291 and appears to have different states, depicted in the Campbell diagram. Hence,  
 292 the assignment of the second bending mode pair does not work reliably. A  
 293 cause for this behaviour could not be found, however, interactions with the  
 294 rotor may be an explanation.

criterion	minimum	maximum	max standard deviation
power in kW	0	-	-
pitch angle in degree	-2	25	2.5
nacelle angle in degree	-	-	0.3

Table 2: Data selection criteria based on 10 minutes aggregated SCADA data.

295 In the following, this study focuses on the dynamics of the plant in operation.  
 296 In order to exclude uncertainties due to transient time signals caused by the  
 297 start-up and shut-down of the wind turbine as far as possible, only data sets  
 298 are considered where the aggregated 10-minute SCADA data indicate constant  
 299 operation. The selection criteria ensuring this are listed in Table 2. The me-  
 300 dians of the identified natural frequencies, as well as the identification rate for  
 301 the selected data are listed in Table 1. The reason for the relatively low identi-  
 302 fication rates is on the one hand that only completely identified bending mode  
 303 pairs are used. In case of strongly unequal excitation of the pair, it may occur  
 304 that only one mode is identified and the identification of the pair is thus incom-  
 305 plete. In addition, harmonic excitation and other transient effects can disturb  
 306 the identification as in the case of the 2nd bending mode. In general, it is often

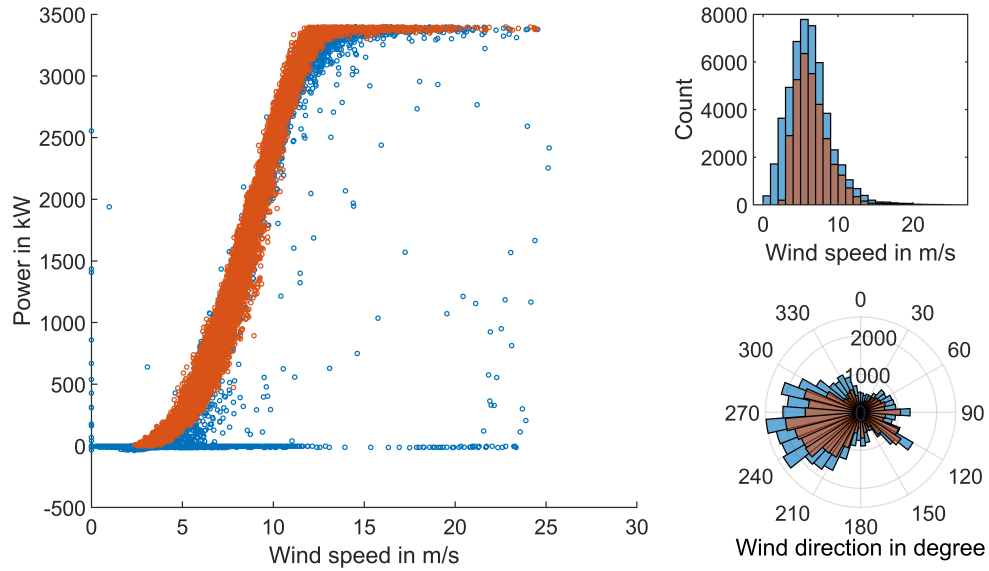


Figure 7: Power curve, wind speed and wind direction distribution of the investigated wind turbine, blue: all data, red: selected data (October 2021-September 2022).

307 the case that some modes are more difficult to identify when monitoring wind  
 308 turbine towers [16], leading to lower identification rates. Figure 7 shows the  
 309 power curve, the distribution of the wind speed and the wind direction of all  
 310 data as well as the selected data during the almost 12 month period considered  
 311 in this study. As stated before, only data sets belonging to operation conditions  
 312 are used, and data points outside the expected power curve are not taken into  
 313 account. Regarding the wind speeds, data sets below 2.7m/s are consistently  
 314 omitted. Otherwise, the distributions of wind speeds and wind direction remain  
 315 qualitatively the same. To exclude inconclusive identification results, identifi-  
 316 cations were not considered if the determined identification uncertainty of the  
 317 natural frequency and damping was detected as an outlier using a Hampel filter  
 318 [25] with a window length of 144, which corresponds to one measurement day.

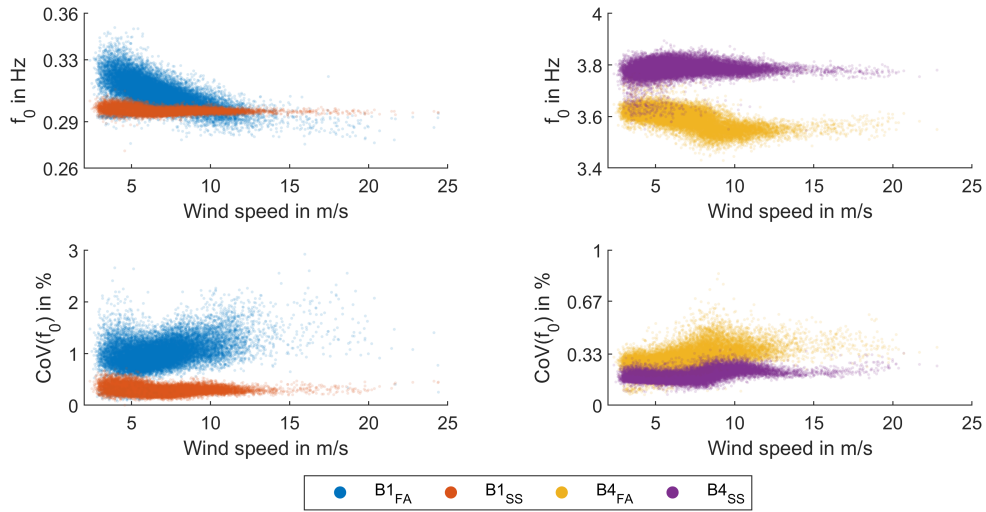


Figure 8: Top: Natural frequencies of the first and fourth bending mode pairs as a function of wind speed. Bottom: Coefficient of variation (CoV) in percent of the natural frequencies as a function of wind speed.

319 For further investigations, the first and fourth bending mode pairs are selected,  
 320 as the first is the closest and the fourth is the best-separated mode pair. The top  
 321 panel of Figure 8 shows the natural frequencies depending on the wind speed.  
 322 The first bending mode in FA direction has a much stronger dependence on  
 323 wind speed than the mode in SS direction. In addition, the observed scattering  
 324 of the FA direction is significantly higher. Similar observations are made for  
 325 the fourth bending mode pair, although the scattering is smaller. The identifi-  
 326 cation uncertainty of the natural frequencies at the bottom of Figure 8 shows  
 327 that the FA direction is identified with a higher uncertainty than the SS direc-  
 328 tion. The main reason for this difference is the aerodynamic damping [16],  
 329 which leads to a significantly higher damping in the FA direction, as shown  
 330 in Figure 9. A higher damping leads to higher identification uncertainties of  
 331 the frequency. Since the damping of the first FA bending mode increases with

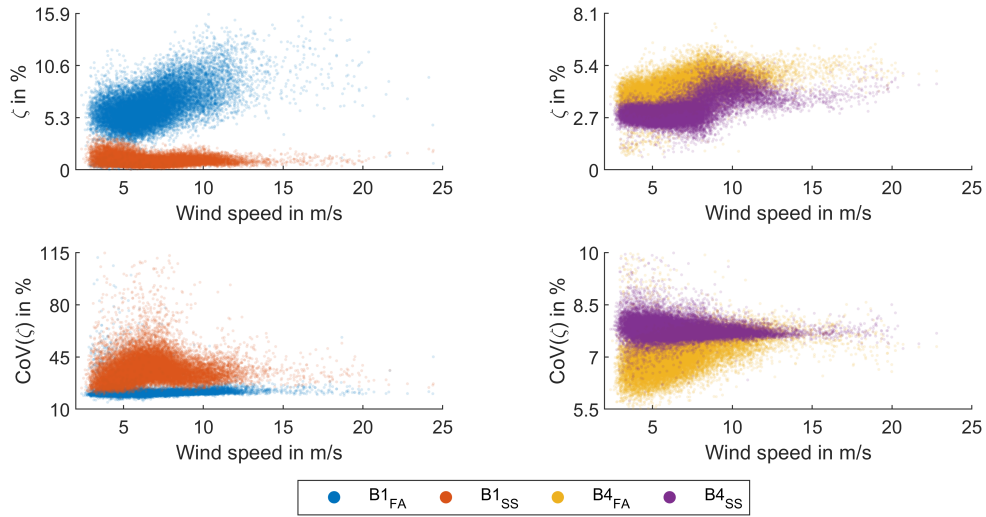


Figure 9: Top: Damping of the first and fourth bending mode pairs as a function of wind speed. Bottom: Coefficient of variation (CoV) in percent of the damping as a function of wind speed.

332 the wind speed, the identification uncertainty of the natural frequency also in-  
 333 creases. For the fourth bending mode pair, the highest damping is determined  
 334 between wind speeds of 4 and 12 m/s, so the corresponding natural frequen-  
 335 cies are determined with the highest uncertainty in this range. In general,  
 336 the identification of the damping is associated with significantly higher uncer-  
 337 tainties than the identification of natural frequencies [20] and this can also be  
 338 confirmed for the data sets used in this study. Regarding the uncertainties of  
 339 the damping identification, in the lower panel of Figure 9 it is noticeable, that  
 340 the damping identification of the SS modes is relatively more uncertain than  
 341 the damping identification of the FA modes. However, the absolute uncertainty  
 342 of the damping of the SS modes is still significantly lower than that of the  
 343 FA modes. In addition, the damping of the fourth bending mode pair can be



344 identified more reliably than the damping of the first mode pair. This is due  
 345 to the length of the 10-minute data sets used. With an increasing number of  
 346 vibration periods in the 10-minute interval, the damping identification becomes  
 347 less uncertain. The uncertainty of the first bending modes in SS direction is  
 348 remarkable, as the uncertainties in identification of the damping are more scat-  
 349 tering at wind speeds below 10 m/s, as opposed to the trends observed for the  
 350 other modes.

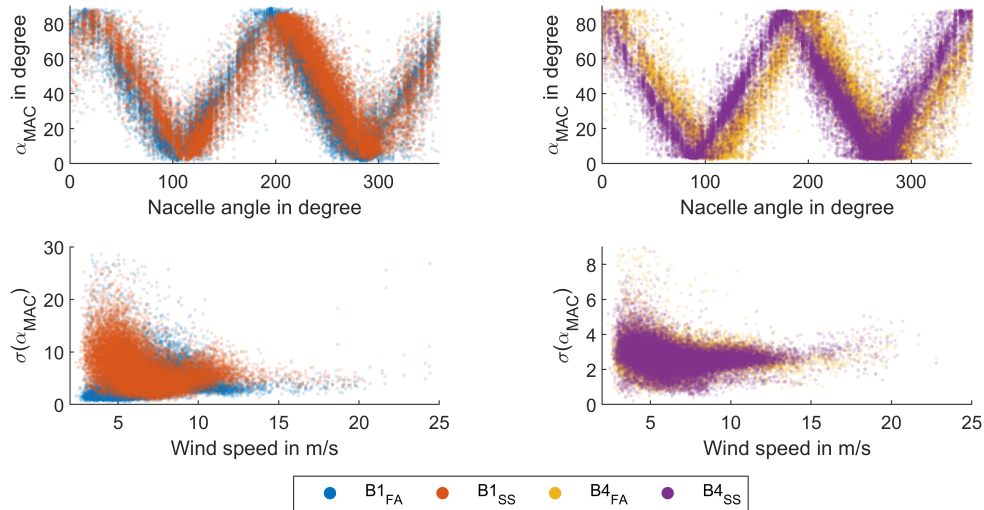


Figure 10: Top:  $\alpha_{MAC}$  of the first and fourth bending mode pairs as a function of the nacelle angle. Bottom: Standard deviation of the  $\alpha_{MAC}$  for the first and fourth bending mode pairs as a function of wind speed.

351 In the context of closely spaced modes, the mode shapes are of particular  
 352 interest. The main uncertainty concerns the alignment of the mode shape in  
 353 the mode subspace. In the case of wind turbine towers, the mode shape also  
 354 changes due to the nacelle position. Therefore, in the top panels of Figure 10,  
 355 the  $\alpha_{MAC}$  for both pairs of modes is shown as a function of the nacelle position.  
 356 The reference modes have been identified in the main wind direction at a

357 nacelle position of  $270^\circ$ , so at that nacelle position, the  $\alpha_{MAC}$  value is close to 0.  
 358 There are few measurement data sets at nacelle positions below  $100^\circ$  and above  
 359  $320^\circ$ , which is due to the wind direction distribution. The deviation between the  
 360 mode pairs depending on the nacelle position that can be observed in Figure  
 361 10 is due to the asymmetric stiffness distribution around the circumference,  
 362 which may result from imperfections in the dry joints between the concrete  
 363 segments or attachments. The uncertainty of the  $\alpha_{MAC}$  is shown as a function  
 364 of the wind speed. The fourth pair of bending modes is already well separated,  
 365 so the uncertainty of the  $\alpha_{MAC}$  is relatively small, which leads to a standard  
 366 deviation of the direction angle of less than  $5^\circ$  for both bending modes. In the  
 367 case of the first bending mode pair, the standard deviation of the  $\alpha_{MAC}$  is very  
 368 high at low wind speeds, especially for the SS mode.

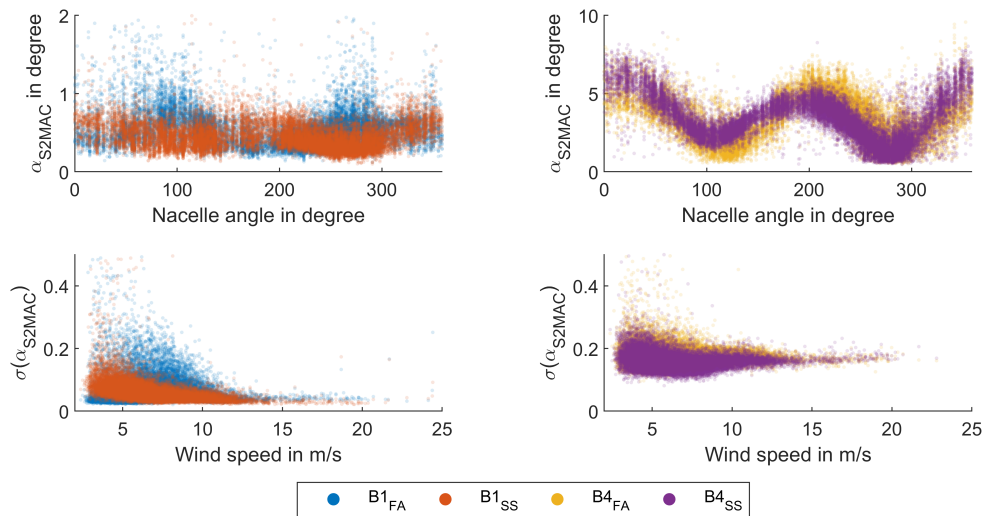


Figure 11: Top:  $\alpha_{S2MAC}$  of the first and fourth bending mode pairs as a function of the nacelle angle. Bottom: Standard deviation of the  $\alpha_{S2MAC}$  for the first and fourth bending mode pairs as a function of wind speed.

369 In contrast, the  $\alpha_{S2MAC}$  depicted in Figure 11 changes less depending on the  
370 nacelle position. In particular, the first bending mode pair appears to have a  
371 relatively constant  $\alpha_{S2MAC}$  regardless of the nacelle position. For the fourth  
372 bending mode pair, there is a clear dependence of the  $\alpha_{S2MAC}$  on the nacelle  
373 position. This indicates that the mode subspace changes slightly as a function  
374 of nacelle position, which is presumably due to an asymmetric stiffness distri-  
375 bution over the circumference. The uncertainties of the  $\alpha_{S2MAC}$  in Figure 11  
376 are much lower than those of the  $\alpha_{MAC}$  in Figure 10 for both bending mode  
377 pairs. This indicates that the  $\alpha_{S2MAC}$  successfully eliminates the alignment un-  
378 certainty. Furthermore, the uncertainty of the  $\alpha_{S2MAC}$  of the first bending mode  
379 pair is significantly lower than that of the fourth one. One reason could be a  
380 better signal to noise ratio, as has already been shown in [19]. In addition, the  
381 mode shape of the first bending mode pair has no nodal points at the sen-  
382 sor points considered, in contrast to the fourth bending mode pair, so that the  
383 measurement noise has a minor influence.

384 The direction angle  $\gamma$  expresses for symmetrical tower structures the align-  
385 ment of the mode shape in the mode subspace [17] and is shown in Figure  
386 12. The uncertainty of  $\gamma$  is higher the closer the modes are. As expected, the  
387 direction angle  $\gamma$  in the top of panel of Figure 12 depends linearly on the na-  
388 celle position. However, a larger scatter can be observed over the whole trend.  
389 This is due to the non-synchronous SCADA and the uncertainty of the direction  
390 angle shown in the bottom panel of Figure 12. The uncertainty is presented  
391 as a function of the wind speed. Considering this result, it is noticeable that

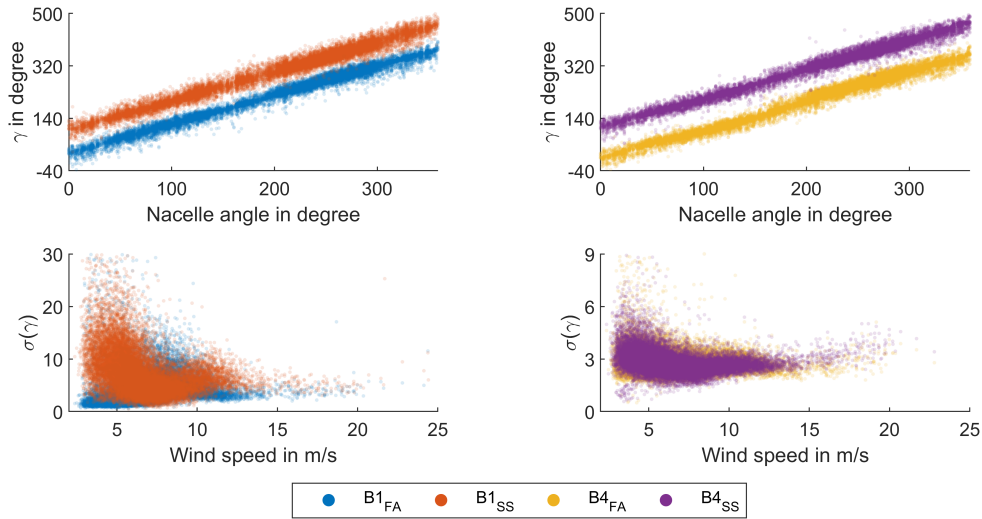


Figure 12: Top: Directional angle  $\gamma$  of the first and fourth bending mode pairs as a function of the nacelle angle. Bottom: Standard deviation of the directional angle  $\gamma$  as a function of wind speed.

392 the uncertainty of the direction angle is similar to the uncertainty of the  $\alpha_{MAC}$   
 393 in Figure 10. This demonstrates clearly that the main uncertainty of the mode  
 394 shapes of bending modes of wind turbine support structures originates from  
 395 the alignment uncertainty within the mode subspace.

396 Throughout this investigation, it must be taken into account that the assump-  
 397 tions of BAYOMA, such as white noise as an excitation source and a linear  
 398 time-invariant system, are violated. Therefore, the calculated uncertainties are  
 399 indicative, but do not exactly correspond to the true uncertainties. However,  
 400 for the practical application, it can be stated that BAYOMA can be used to ob-  
 401 tain consistent dynamical identifications of the turbine structure of an onshore  
 402 wind turbine.

403 Based on this investigation, it can be concluded that the  $\alpha_{MAC}$  of the tower bend-

404 ing mode shapes with their high identification uncertainties independent of the  
405 nacelle position cannot serve as a reliable monitoring parameter. Instead, the  
406 identified mode shapes should be compared with a mode subspace using the  
407  $\alpha_{S2MAC}$ . This eliminates the high alignment uncertainty. As known from other  
408 studies [16], the more weakly damped SS natural frequencies can be identified  
409 more reliably than the FA natural frequencies. Nevertheless it is recommend-  
410 able to also consider both natural frequencies as monitoring parameters, like  
411 [8, 16, 26].

#### 412 **4. Summary and Outlook**

413 In this study, the applicability of BAYOMA to identify closely spaced bending  
414 modes from a tower of an onshore wind turbine in operation was investigated.  
415 The identification and the corresponding uncertainties provided plausible re-  
416 sults despite the presence of harmonic excitation from the rotor. More strongly  
417 damped natural frequencies are much more uncertain to identify. Conse-  
418 quently, the less damped natural frequencies in SS direction can clearly be  
419 more reliably identified than the ones in FA direction. As typical for structures  
420 exhibiting closely spaced modes, the mode shapes can only be identified with  
421 high uncertainty, because the alignment of the mode in the mode subspace  
422 is very uncertain. Therefore, the  $\alpha_{MAC}$  as well as the mode alignment angle  
423 are not suitable as reliable monitoring parameters. This does not apply to the  
424  $\alpha_{S2MAC}$ , which proved to be a reliable monitoring parameter, as already shown  
425 in previous studies for tower structures [19, 17].

426 Several future research approaches result from this study. In the onshore hy-  
427 brid steel and concrete tower investigated, harmonic excitation did not have  
428 a significant impact, so that the identification of the modal parameters with  
429 BAYOMA worked well. For a more general statement it is thus necessary to  
430 investigate how harmonic excitation can affect the modal identification of wind  
431 turbine support structures constructed exclusively from steel, both onshore  
432 and offshore. The examination of the modal parameters clearly showed that  
433 they vary due to EOCs, so that the next step is to normalise the data for a  
434 reliable SHM-scheme. The uncertainties of the modal parameters indicate het-  
435 eroscedasticity with respect to the EOC, i.e. a variability in dependence of the  
436 EOCs. Therefore, heteroscedastic Gaussian processes might be a good method  
437 for data normalisation to map this variability.

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448 **6. Author contribution**

449 Conceptualisation: C.J. Methodology: C.J. Formal analysis: C.J.; S.M. Investiga-  
450 tion: C.J. Writing Original Draft: C.J. Visualisation: C.J.; L.L. Supervision: T.G.;  
451 R.R. Writing Review and Editing: S.M.; L.L.; B.H.; T.G.; R.R. Resources: R.R.  
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